A Refrigerated Dynamic Seal

R. C. Hendricks

Lewis Research Center

Cleveland, Ohio

Prepared for the Cryogenic Engineering Conference/ International Cryogenic Materials Conference Colorado Springs, Colorado, August 15-19, 1983



A REFRIGERATED DYNAMIC SEAL

R. C. Hendricks

National Aeronautics and Space Administration Lewis Research Center Cleveland Ohio 44135 USA

ABSTRACT

A new sealing concept is proposed where the fluid to be sealed flows through a refrigerated housing or constriction, is frozen to the housing during the transient phase, and under steady state conditions provides a dynamic seal with essentially 'zero' leakage.

INTRODUCTION

In all applications the primary purpose of a seal is to prevent the loss of a fluid, i.e., minimize leakage. In turbomachine applications there are usually several seals between bearing supports. In many instances, improper design of these seals have been the cause of failures and derating of turbomachines, reference 1. Thus a secondary purpose for a seal, and often equally important as leakage control, is their use in prevention of dynamic instabilities.

In some applications one fluid may be sufficiently cooler than another and could serve as a heat sink for the fluid within the seal either by direct heat exchange, a heat pipe, or other such refrigeration devices. In these cases, the seal leakage and dynamic parameters as stiffness and dampening can be actively controlled by either the formation of a solid within the seal aperture or increasing the viscosity or both. The rotating member generates energy through viscous dissipation which is then balanced by the refrigeration load cited above.

In this paper a type of self sealing configuration is proposed for applications to turbomachines using conventional or cryogenic refrigeration technology to control the viscosity and solid plug buildup within the seal aperture.

ANALYSIS

Consider a shaft seal where either the working fluid can solidify or a solid plug can be formed through secondary fluid injection. Further consider a seal configuration such that the clearance to radius ratio c/R = (b - t)/R << 1, so that the problem can then be addressed as one-dimensional; refrigera-

^{*} a. While infinite at the critical point, viscosity has a minimum in the thermodynamic critical region.

b. At low temperatures (near the triple point) viscosity has a maximum for quantum fluids.

c. The temperature effect is usually strong for liquids (exp (a/T)) and weak for gases, $T^{2/3}$. Further for a synthetic oil such as Mobile RL-714 Stock 509, a temperature change from 60° to 20° C (140° to 68° F), increases the viscosity by a factor of 4, reference 3.

d. One now must consider the stability of operating in the critical region with bearings etc. as $\boldsymbol{\eta}$ has minimum.

For gaseous nitrogen refrigerant flowing at 3 meters per second (10 ft/s) through a 6.4 millimeter (1/4 in.) diameter tube, the Reynolds number (Re) is 7800 and h_a is 28 watts per square meter (9 B/hr-ft²-F). On the water side, for a leakage velocity of 1.8 meters per second (6 ft/s) with a hydraulic diameter of 2.5 millimeter (0.1 in.) the Reynolds number becomes 9700 and h_i is 50 watts per square meter (16 B/hr-ft²-F).

Using these values with b of 1.2 millimeter (0.05 in.) (i.e., t = 0), would require two minutes for an ice plug to form; for b of 0.75 millimeter (0.03 in.) would require less than one minute. The actual 'pinch off' clearance (b - t) is self limiting; for as soon as the clearance becomes very small, S becomes large and melts the ice. An additional feature is that the maximum temperature occurs within the aperture channel and not at the rotating shaft thereby preventing complete closure so long as the shaft is rotating.

EXPERIMENTAL RESULTS

An apparatus, shown in figure 3, was constructed to illustrate the principle of a self sealing system and to investigate its effectiveness for practical applications. The fluid to be sealed was water and nitrogen was the refrigerant. Apertures of the orifice and Borda types (ref. 4) were used as shaft seals and placed between bearing supports. With the shaft rotating, water was added to the reservior which leaked profusely, figure 4. As the nitrogen cooled an annular region around each seal the leakage stopped entirely, figure 5. The time required to stabilize the rotor and seal the system is generally shorter for the Borda type seal as would be expected due to better heat transfer and larger contact area. For eccentric seals or those rubbed or misaligned or badly damaged, the solid plug (ice layer) builds up quickly, and with sufficient refrigeration the leak stops completely.

For this system with 51 millimeters (2 in.) of water head at 12° C, the leak rate was 21 milliliters per second through the Borda aperture and 33 milliliters per second through the orifice aperature. At refrigerant loads greater than 200 watts, the time required to stop the leakage was less than 25 seconds for either aperture and so the simplified model appears conservative. The threshold refrigerant loads were found to be 80 and 170 watts for the Borda and orifice respectively.

Once the plug is formed and maintained, the seal should not 'wear' and should adjust to system perturbations.

POSSIBLE APPLICATIONS AND LIMITATIONS

- 1. The method can be applied to systems requiring very precise seal apertures. During the assembly and 'rub in' phase, a temperature gradient is imposed across some suitable seal material such that the interface is raised to a temperature $T_m + \Delta T$ where T_m is the melting temperature and $T_m > T_{0,max}$. Rotate the shaft system slowly and increase to operating speed; then bring the temperature back to equilibrium such that $T < T_m$ while checking torque and leakage. If not satisfactory, then repeat. Such a seal can be within microns of the rotating member and limited by the characteristic surface roughness (ref. 5).
- 2. A portable system with a flexible coolant system as a heat pipe or flex tube could be used as a quick fix to arrest leaking seals of operating systems until such time as it is shutdown for repairs. The concepts could

also be applied to a dual-spray system whereby a matrix material capable of 'holding' the refrigerant is sprayed onto a leak form one nozzle and the refrigerant from another.

- 3. For the dynamic applications, a rub will increase the amount of fluid within the passage. Continuity will require more fluid to be flow circum-ferentially and provide damping of the system. This aspect of enhanced stability can be as important as leakage control.
- 4. The maintenance of a thin fluid layer between the solid plug and the rotating member can significantly reduce the friction over that of dry sliding contact type seals.
- 5. It is possible to use the method in reverse, i.e., add heat to melt a region near the rotating member to facilitate less energy loss due to solid contact during rotation, an ablative method, reference 6.
- 6. This method can not be used as a static seal unless auxillary heat is available to melt a thin layer adjacent to the rotating member prior to startup.
- 7. The method can not be used to form a plug when the fluids are noncondensibles; however, the viscous effect on axial pressure drop can still be significant. Secondary fluid injection could be used to overcome this difficulty.
- 8. Due to the concentration effects of sudden ice formation, corrosion of the shaft material may occur.

SUMMARY

A new sealing concept is proposed where the fluid to be sealed flows through a refrigerated housing or constriction. As the fluid cools the viscosity increases and the resistance to leakage increases and in some cases leakage stops completely. In some cases this may be sufficient to establish the desired leakage rates and in others the temperature needs to be lowered below the freezing point of the fluid in the aperture to build and maintain a solid plug. Under steady operating conditions the viscous dissipation provided by the rotating member maintains a thin liquid film that can sustain large axial pressure gradients thereby providing a dynamic seal with essentially 'zero' leakage.

REFERENCES

- 1. Rotordynamic Instability Problems in High-Performance Turbomachinery. NASA CP-2133, 1980.
- 2. Rohsenow, W. M., and Hartnett, J. P.: eds.: Handbook of Heat Transfer. McGraw-Hill, 1973.
- 3. Braun, M.J.; and Hendricks, R.C.: An Experimental Investigation of Vapor-ous/Gaseous Cavity Characteristics of an Eccentric Journal Bearing. ASLE Preprint 82-LC-3A-1, Oct. 1982.
- 4. Hendricks, R.C.; and Stetz, T. T.: Flow through Axially Aligned Sequential Aperatures of the Orifice and Borda Types. ASME paper 81-HT-79, Aug. 1981.
- Hendricks, R. C.: (NASA patent disclosure on preferential surface roughness in seals)
- 6. Hendricks, R. C.: (NASA patent disclosure on uses of ablative materials in seals)

SYMBOLS

```
b
    aperture static clearance
    aperture dynamic clearance
С
   specific heat
    hydraulic diameter
d
    mass flux
G
    latent heat
Н
    heat transfer coefficient
h
    thermal conductivity
k
    ratio of heat transported by refrigerant to that conducted through the
    solid plug, eq. 5.
    ratio of heat input form the seal to that absorbed by the refrigerant,
    eq. 4.
Nu Nusselt number, hd/k
Pr Prandtl number, Cp \eta/k
R
    shaft radius
Re Reynolds number, G d/n
    viscous dissipation
S
T
    temperature
t
    time
    ( ) increment
Δ
    density
    dimensionless time in terms of convection, conduction and latent energies
     at the interface, eq. 3.
θ
    time
    viscosity
η
    angular velocity
```

Subscripts

- a coolant side
 f fusion or solid
 i seal fluid side
 m melting
 max maximum
- o reference or stagnation

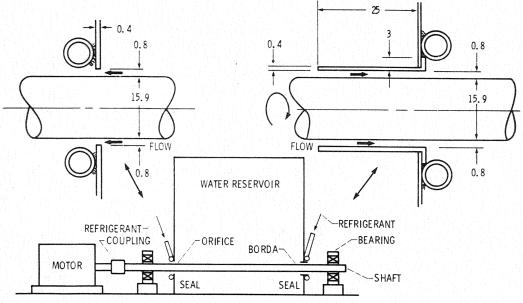


Figure 3. - Schematic of the experimental test apparatus for Borda and orifice type refrigerated seals. Dimensions in mm.

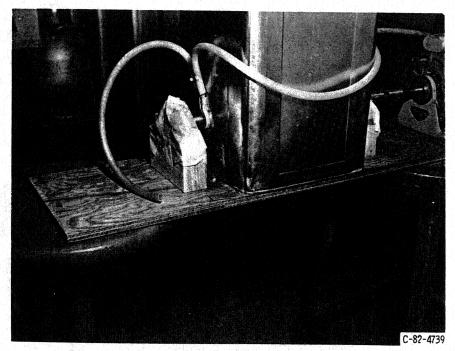


Figure 4. - Photograph of the non-refrigerated seal apparatus with rotation.

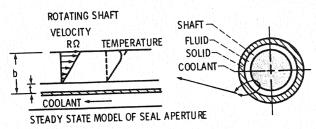


Figure 1. - Schematic of velocity and temperature profiles for a refrigerated seal configuration.

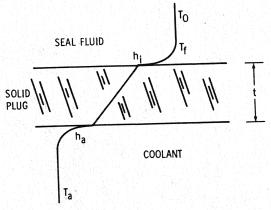


Figure 2. - Temperature profile for the refrigerated seal model.

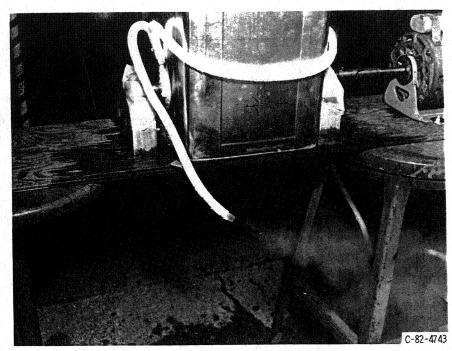


Figure 5. - Photograph of the apparatus after application of refrigeration, with rotation.

			<u> </u>
1. Report No. NASA TM-83378	2. Government Accession	on No.	3. Recipient's Catalog No.
4. Title and Subtitle			5. Report Date
A REFRIGERATED DYNAMIC SEAL			
			6. Performing Organization Code 505-32-52
7. Author(s)			8. Performing Organization Report No.
R. C. Hendricks			E-1649
			10. Work Unit No.
9. Performing Organization Name and Address			
National Aeronautics and Space Administration Lewis Research Center Cleveland, Ohio 44135			11. Contract or Grant No.
			13. Type of Report and Period Covered
2. Sponsoring Agency Name and Address			Technical Memorandum
National Aeronautics and S	Space Administration		14. Sponsoring Agency Code
Washington, D. C. 20546			The Sponsoring Agency Code
Conference, Colorado Spri	ings, Colorado, Augus	τ 15-19, 1983.	
	frozen to the housing	during the transi	ows through a refrigerated ient phase, and under steady eakage.
A new sealing concept is p housing or constriction, is	frozen to the housing	during the transi	ient phase, and under steady
A new sealing concept is p housing or constriction, is state conditions provides a	frozen to the housing	during the transi	ient phase, and under steady
A new sealing concept is p housing or constriction, is	s frozen to the housing a dynamic seal with es	during the transi	ent phase, and under steady eakage.
A new sealing concept is p housing or constriction, is state conditions provides a	s frozen to the housing a dynamic seal with es	during the transisentially 'zero' lo 18. Distribution Stateme Unclassified - STAR Categor	ent phase, and under steady eakage.